

ABSTRACT

The present invention provides a turbine moving blade, a turbine stationary blade, and a turbine split ring which are capable of restraining
5 the deterioration and peeling-off of a thermal barrier coating easily and surely, and a gas turbine capable of enhancing the energy efficiency by increasing the temperature of combustion gas.

The turbine moving blade provided in a turbine
10 constituting the gas turbine includes a platform having a gas path surface extending in the combustion gas flow direction, and a blade portion erecting on the platform. The thermal barrier coating covering the gas path surface is formed so
15 as to go around from the gas path surface to an upstream-side end face and a downstream-side end face of the outer peripheral faces of the platform.